Risk Assessment

A complicated system, such as a spacecraft has many things that could make the mission a failure. To reduce these tendencies and know where possible problem areas may happen and how to best eliminate their occurrences a Risk Assessment table was constructed. This can be seen in Figure 1 and Table 1.

When considering a complicated system such as a spacecraft there are always things that can go wrong. Do get a better handle on what could possibly go wrong and potential ways to make sure it doesn’t a Risk Assessment table was constructed. This can be seen in Figure 1 and Table 1.

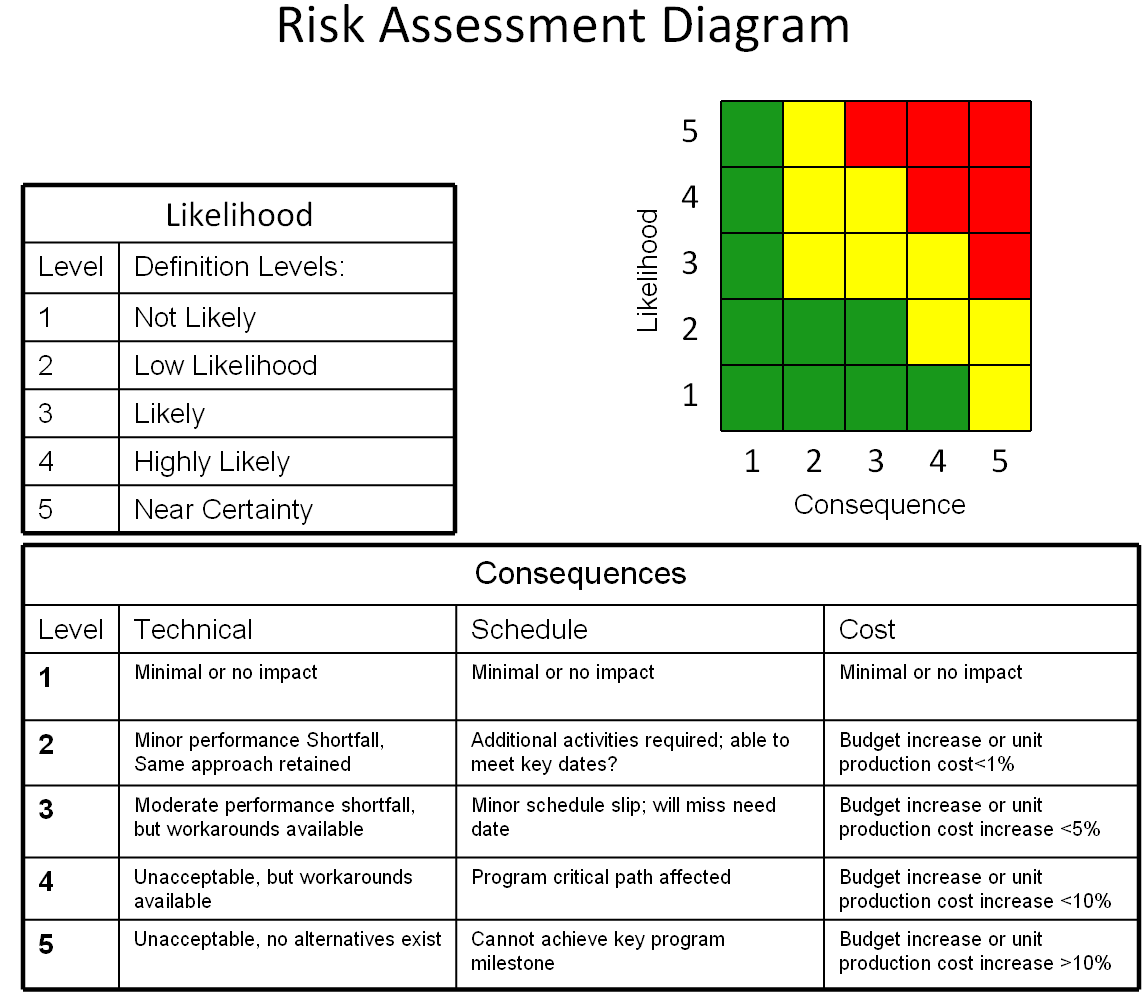
Figure 1: Risk assessment layout

Table 1: Risk and the likelihood and consequence of them occurring

|  |  |  |  |
| --- | --- | --- | --- |
|  | **Risk** | **Likelihood** | **Consequences** |
| 1 | Loss of Communication | 2 | 5 |
| 2 | Collision between spacecraft | 3 | 3 |
| 3 | The spacecraft did not separate properly | 3 | 2 |
| 4 | Spacecraft unable to locate each other | 3 | 4 |
| 5 | Unable to generate enough power | 2 | 3 |
| 6 | Spacecraft separate too fast from each other | 3 | 3 |
| 7 | The spacecraft end up too far apart from each other | 3 | 4 |
| 8 | Run out of propellant before end of mission | 1 | 5 |
| 9 | An error occurs during the integration of spacecraft components | 2 | 4 |
| 10 | Deorbits before six months | 1 | 5 |
| 11 | A spacecraft component fails on orbit | 1 | 3 |
| 12 | A leak develops on the propulsion unit | 1 | 3 |
| 13 | A failure occurs on the satellite during launch | 1 | 3 |
| 14 | Solar panels malfunction | 2 | 3 |
| 15 | Spacecraft components operate out of operational temperature range | 1 | 2 |
| 16 | Radiation damage | 1 | 2 |

|  |  |  |  |  |  |
| --- | --- | --- | --- | --- | --- |
| 5 |  |  |  |  |  |
| 4 |  |  |  |  |  |
| 3 |  | 3 | 2 5 6 | 4 7 |  |
| 2 |  |  | 14 | 9 | 1 |
| 1 |  | 15 16 | 11 12 13 |  | 8 10 |
|  | 1 | 2 | 3 | 4 | 5 |

Figure 2: Risk Assessment matrix

Loss of communication:

Upon reaching orbit, if we were to lose communication with the spacecraft this would have catastrophic results. With no communication with the spacecraft this would mean we could not deliver commands instructing the vessel what it was supposed to do, thus killing the mission. To potentially mitigate this ground testing would be performed on what could go wrong such as a break from the vibration of the rocket, failure to deploy the antennas (if necessary), lack of strength of the radio, antennas could be miss directed. To test these possibilities we would conduct: vibration tests to make sure nothing vibrates loose, burn circuit tests to make sure things deploy correctly (if necessary), Long range of sight testing, and ensure magnets are placed directly.

When finally reaching orbit, communication could be a problem. If something happened to the space craft and we could never talk to it or if we had communication then lost it the mission would have catastrophic results. If there was no communication to the spacecraft no commands could be sent to tell it what to do and the mission would cease. To potentially mitigate this ground testing would be performed on what could possibly go wrong such as a break from the vibration of the rocket, deploy failure of the antennas (if necessary), lack of strength of the radio, and miss directed antennas. To test these: vibrations tests to make sure nothing vibrates loose, burn circuit tests to make sure things deploy correctly (if necessary), Long range line of sight testing, and ensuring magnets are placed correctly.

Collision between spacecraft:

When in orbit it is important to make sure a collision between Nephrite and Jade does not occur. Collision can cause potentially catastrophic damage to the spacecraft. To prevent a collision extensive testing will be performed considering the control of our spacecraft through a 2-axis low friction table or a weightless flight on the “vomit comet”.

When in orbit it is important to make sure a collision with ourselves does not happen. Collision can cause potentially catastrophic damage to the spacecraft. To prevent this extensive testing will be performed considering control of our spacecraft through a 2-axis low friction table or a weightless flight on the vomit comet.

The spacecraft did not separate properly:

If the spacecraft does not separate properly this could result in failure of our mission, as it could not be performed if they are stuck latched together. To mitigate this ground testing will be done to make sure separation mechanisms are working properly. Additional testing can be done on the 2-axis low friction table or a weightless flight on the “vomit comet”.

If the spacecraft does not separate properly this could result in catastrophic failures in our mission, as it could not be performed if they are latched together. To mitigate this ground testing will be done to make sure separation mechanisms are working properly. Additional testing can be done on the 2-axis low friction table or a weightless flight on the vomit comet.

Spacecraft unable to locate each other:

If while in space after separation occurs the spacecraft are unable to locate each other mission status could not be confirmed and thus parts of the mission may be deemed a failure. To help mitigate this location testing will be performed on the ground. Numerous tests will be performed with each spacecraft to ensure there is successful recognition between them.

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Unable to generate enough power:

If our power system browns out or does not have enough power to transmit we have no way of confirming if the mission was completed or not. Testing can be performed on panels by using a constructed “SUN” consisting of numerous full spectrum light bulbs that can replicate the suns conditions in space. Additional testing can be performed in the full function flight test with a completely integrated spacecraft with the same method. This will confirm correct power draw and use.

If our power system browns out or doesn’t have enough power to transmit we have no way of confirming if the mission was completed or not. Testing can be performed on panels by using a constructed “SUN” consisting of numerous full spectrum light bulbs that can replicate the sun conditions in space. Additional testing can be performed in the full function flight test with a completely integrated spacecraft with the same method. This will confirm correct power draw and use.

Spacecraft separate too fast from each other:

If when separation occurs, the spacecraft separates to quickly it would make it extremely difficult to stabilize the relative velocity. This can result in the spacecraft being far enough apart that it would make it difficult to recover the mission status. To ensure this does not happen testing on the 2-Axis low friction table will be performed.

If when separation occurs, the spacecraft separates to quickly it would make it extremely difficult to stabilize relative velocity. This can result in the spacecraft being km apart and mission status being unrecoverable. To ensure this doesn’t happen testing on the 2-Axis low friction table will be performed.

Run out of propellant before end of mission:

If during operation the system runs out of propellant the mission would cease to persist. Depending on the state of the mission this could cause the mission to be incomplete. The most efficient way of mitigating this is to run calculations that state how much propellant we would need.

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An error occurs during the integration of spacecraft components:

Integration errors can occur and depending on which could present different consequences, causing different severities to the mission. The best way we could go about nulling these effects are to have ample documents and practice integrating the structure, confirming successful working conditions after each practice run.

Depending on the severity of the integration error different consequence present itself, causing different severities to the mission. The best we to null the effects are to have ample documents and practice integrating the structure, confirming successful working conditions after each practice run.

Deorbits before six months:  
 Depending on when the spacecraft deorbits depends on the severity on the premature deorbit. This can be prevented by running extensive simulations on STK to determine the orbit analysis.

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A spacecraft component fails on orbit:

Based on which components were to fail during orbit determines the severity of the consequences. Certain things may be out of our control however best way to mitigate this is to have an engineering unit which has all of the parts and which we test but an additional flight unit so the components have a less chance of breaking.

Depending on what fails during orbit determines the severity of the consequence. Something’s may be out of our control but the best way to mitigate this is to have an engineering unit which has all of the parts and which we test but an additional flight unit so the components have a less chance of breaking.

A leak develops on the propulsion unit:

At which point during the mission a leak from the propulsion unit can determine the severity of consequences to the system. The best way to mitigate this is to test the propulsion system severely to ensure successful design.

Depending on the time the leak develops and the severity, this indicates the severity of the consequence to the system. The best way to mitigate this is to test the propulsion system severely to ensure successful design.

A failure occurs on the satellite during launch:

During launch which system were to fail, determines the severity of the consequences. The best way to mitigate this is to put the satellite through vibrational testing as required by NASA to ensure the vibrations won’t tear the spacecraft apart.

Depending on what fails during lunch determines the severity of the consequence. The best way to mitigate this is to put the satellite through vibrational testing as required by NASA to ensure the vibrations won’t tear the spacecraft apart.

Solar panels malfunction:

Power budget could be affected and lessened based on if and how many of our solar panels were to fail. This increasing our time frame between objectives for the satellite as it would require more time to charge the cells and a brown out does not occur. To mitigate this, vibrational testing along with extensive functional testing can prevent panels from breaking.

Depending on how many solar panels fail make our power budget less and we would have to wait longer in between objectives for the satellite to charge and a brown out does not occur. To mitigate this, vibrational testing along with extensive functional testing can prevent panels from breaking.

Spacecraft components operate out of operational temperature range:

Operational temperature ranges of the spacecraft and which of the components were to operate out of the correct range could differentiate severities. To mitigate this, thermal tests will be performed on the spacecraft then a functional test performed afterword to ensure everything is working properly.

Depending on the part out of range this could differentiate severities. To mitigate this, thermal tests will be performed on the spacecraft then a functional test performed afterword to ensure everything is working properly.

Radiation damage:

Having a solar flare or components that cease to work under radiation would cause different severities of mission failure. To mitigate this only radiation-hardened component’s will be used.

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